MATH351201

This question paper consists of 4 printed pages, each of which is identified by the reference **MATH3512**.

Only approved basic scientific calculators may be used

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Examination for the Module MATH3512 (May/June 2005)

Viscous Flow

Time allowed: 3 hours

Do not attempt more than 4 questions. All questions carry equal weight.

1. Viscous fluid is in steady, axi-symmetric motion in the annular region between two infinitely long, concentric cylinders r=a and r=b, where a < b. The flow results from an applied constant pressure gradient -P in the z-direction. Show that the Navier-Stokes equations reduce to

$$\frac{d^2w}{dr^2} + \frac{1}{r}\frac{dw}{dr} = -\frac{P}{\mu}, \qquad u \equiv 0,$$

where z measures distance along the cylinders and r is the radial co-ordinate. u and w are the velocity components in the r and z directions respectively and μ is the coefficient of viscosity.

State the boundary conditions on w(r) and hence find w(r).

Show that the volume flux Q, given by $Q = 2\pi \int_a^b w(r) r dr$, is

$$Q = \frac{P\pi}{8\mu} (b^2 - a^2) \left[b^2 + a^2 - \frac{(b^2 - a^2)}{\log(b/a)} \right].$$

2. Viscous fluid is in slow, steady two-dimensional motion in the region $-\alpha \leq \theta \leq \alpha$ between two fixed plates along $\theta = -\alpha$ and $\theta = \alpha$. The equation governing this slow flow is

$$\nabla^2(\nabla^2\psi) = 0,$$

where ψ is the stream function defined so that the velocity components (u, v) in the r and θ directions are given by

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$$u = \frac{1}{r} \frac{\partial \psi}{\partial \theta}, \qquad v = -\frac{\partial \psi}{\partial r}$$

and where

$$\nabla^2 \equiv \frac{\partial^2}{\partial r^2} + \frac{1}{r} \frac{\partial}{\partial r} + \frac{1}{r^2} \frac{\partial^2}{\partial \theta^2}.$$

Show that the stream function $\psi(r,\theta)$ for this slow flow in the corner can be found in the form $\psi = r^{\lambda} f(\theta)$ provided that the constant λ satisfies the equation

$$\lambda \tan (\lambda \alpha) = (\lambda - 2) \tan [(\lambda - 2)\alpha]$$

It can be assumed that ψ is an even function of θ .

In the special case $\alpha = \pi$, find the smallest positive value for λ that satisfies the equation.

3. Write down the equations for steady, two-dimensional thin film flow (lubrication approximation) of a viscous fluid between two surfaces z = 0 and z = h(x), stating the assumptions made in deriving these equations.

The upper boundary, z = h(x), is fixed. The lower surface, z = 0, is moving with a constant velocity U_0 along its length. Use the thin film equations to show that the pressure p(x) within the gap between the two surfaces is given by the equation

$$\frac{d}{dx}\left(h^3\frac{dp}{dx}\right) - 6\mu U_0\frac{dh}{dx} = 0,$$

where μ is the (constant) coefficient of viscosity of the fluid.

Integrate this equation to obtain an equation for $\frac{dp}{dx}$ in terms of h(x).

The upper surface has a finite length 2L with the pressure having a constant value p_0 outside this region. For an upper surface given by $h(x) = \frac{h_0}{L}(2L+x)$ for $-L \le x \le L$, show that the pressure p(x) is given by

$$p(x) - p_0 = -\frac{3\mu U_0 L(L^2 - x^2)}{2h_0^2 (2L + x)^2}.$$

4. Consider the steady flow U_0 of a viscous fluid, with kinematic viscosity ν , past a body of typical size L. Define the Reynolds number Re for this flow and discuss briefly the significance of Re in the flow of a viscous fluid.

Explain why it is necessary to introduce a boundary layer to determine the flow past a solid body at large Reynolds number. Use a scaling argument to estimate the thickness of this boundary-layer region in terms of the Reynolds number.

Show that, for steady, two-dimensional boundary-layer flow, the vorticity $\zeta = \frac{\partial u}{\partial y}$ satisfies the equation

$$\mathbf{u} \cdot \nabla \zeta = \nu \frac{\partial^2 \zeta}{\partial y^2}, \quad \text{where } \mathbf{u} = (u, v).$$

Explain the significance of the terms in this equation.

Describe what is meant by boundary-layer separation.

Use the boundary-layer equations to show that the skin friction τ_w has a singularity at separation $(x = x_s)$ of the form

$$\tau_w \propto (x_s - x)^{1/2}$$
 as $x \to x_s^-$.

What is the significance of this singularity for solutions of the boundary-layer equations?

5. The boundary-layer equations for steady, two-dimensional flow near a solid boundary (y = 0) are

$$\begin{split} \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} &= 0, \\ u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} &= U \frac{dU}{dx} + v \frac{\partial^2 u}{\partial y^2}, \end{split}$$

where U(x) is the outer flow and (u, v) are the velocity components in the x and y directions respectively.

Show that, for an outer flow $U(x) = a x^m$ (a and m constants), the boundary-layer equations have a similarity solution in the form

$$\psi = (a\nu)^{1/2} x^{(m+1)/2} f(\eta), \quad \text{with} \quad \eta = \left(\frac{a}{\nu}\right)^{1/2} x^{(m-1)/2} y,$$

where ψ is the stream function, defined in the usual way, and where $f(\eta)$ satisfies the equation

$$f''' + (\frac{m+1}{2})ff'' + m(1 - f'^2) = 0,$$

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where primes represent differentiation with respect to η .

Show that the further change of variables

$$f = \left(\frac{2}{m+1}\right)^{1/2} F, \qquad \overline{\eta} = \left(\frac{m+1}{2}\right)^{1/2} \eta$$

transforms the equation into

$$F''' + FF'' + \beta(1 - F'^2) = 0$$
 with $\beta = \frac{2m}{m+1}$,

where primes now represent differentiation with respect to $\overline{\eta}$. What are the boundary conditions satisfied by $F(\overline{\eta})$?

To determine how the solution to this equation behaves for β large, we make the transformation

$$F = \beta^{-1/2} \, \phi, \qquad \zeta = \beta^{1/2} \, \overline{\eta}.$$

Show that the equation at leading order for β large is then

$$\phi''' + 1 - \phi'^2 = 0$$

(where primes now represent differentiation with respect to ζ) and that

$$\phi' = 3 \tanh^2(\frac{\zeta}{\sqrt{2}} + \gamma) - 2$$
 (where $\tanh^2 \gamma = 2/3$)

satisfies the equation and the required boundary conditions.

END