## MATH351201

This question paper consists of 4 printed pages, each of which is identified by the reference **MATH3512**.

Only approved basic scientific calculators may be used

## ©UNIVERSITY OF LEEDS

Examination for the Module MATH3512 (May/June 2003)

## Viscous Flow

Time allowed: 3 hours

Do not attempt more than 4 questions. All questions carry equal weight.

1. Viscous fluid is in steady motion in a channel between two infinite fixed plates at a constant distance h apart. The flow results from an applied constant pressure gradient -P and by fluid being withdrawn transversally through the plate y=0 with a constant speed  $V_w$  and being injected through the plate y=h at the same rate. Show that the Navier-Stokes equations reduce to

$$\frac{d^2u}{dy^2} + \frac{V_w}{\nu} \frac{du}{dy} = -\frac{P}{\mu}, \qquad v \equiv -V_w$$

for this problem. (x and y measure distance along the channel and normal to it and u, v are the velocity components in the x and y directions respectively.)

Solve this equation, subject to the appropriate boundary conditions, to find u(y).

Show that, by making the transformation

$$u = \frac{Ph}{\rho V_w} U, \qquad Y = \frac{y}{h},$$

the equation becomes in dimensionless form

$$\frac{1}{R}\frac{d^2U}{dY^2} + \frac{dU}{dY} = -1$$

where  $R = \frac{V_w h}{\nu}$  is the Reynolds number. What are the boundary conditions? Show that this is a singular problem in the limit as  $R \to \infty$ , with a leading-order solution U = 1 - Y in the outer region. Find the scaling for Y in the boundary layer (inner solution) near Y = 0. Hence determine the leading-order term for U in this region.

2. Viscous fluid occupies the region  $0 < \theta < \alpha$ . The fluid is in slow, steady two-dimensional motion caused by a plate along  $\theta = 0$  moving with a constant velocity  $U_0$  in a direction along its length and  $\theta = \alpha$  is a free surface on which the shear stress is zero.

The equation governing this slow flow is

$$\nabla^2(\nabla^2\psi) = 0,$$

where  $\psi$  is the stream function defined so that the velocity components (u, v) in the r and  $\theta$  directions are given by

$$u = \frac{1}{r} \frac{\partial \psi}{\partial \theta}, \quad v = -\frac{\partial \psi}{\partial r}$$

and where

$$\nabla^2 \equiv \frac{\partial^2}{\partial r^2} + \frac{1}{r} \frac{\partial}{\partial r} + \frac{1}{r^2} \frac{\partial^2}{\partial \theta^2}.$$

Given that the shear stress  $p_{r,\theta}$  in polar co-ordinates is given by

$$p_{r,\theta} = \mu \left( r \frac{\partial}{\partial r} \left( \frac{v}{r} \right) + \frac{1}{r} \frac{\partial u}{\partial \theta} \right),$$

show that this leads to the boundary conditions

$$\psi = 0,$$
  $\frac{\partial^2 \psi}{\partial \theta^2} = 0$  on  $\theta = \alpha$ .

What are the boundary conditions on  $\theta = 0$ ?

By looking for a solution in the form

$$\psi(r,\theta) = U_0 \, rf(\theta),$$

determine  $\psi$ .

Use your result to show that the velocity  $u_s$  on the free surface is

$$u_s = U_0 \left( \frac{\alpha \cos \alpha - \sin \alpha}{\alpha - \sin \alpha \cos \alpha} \right).$$

3. Write down the equations for steady, two-dimensional thin film flow (lubrication approximation) of a viscous fluid on a fixed surface z = 0, stating the assumptions made in deriving these equations.

The upper boundary, z = h(x), is moving with a constant velocity  $U_0$  along its length. Use your thin film equations to show that the pressure p within the gap betwen the two surfaces is given by the equation

$$\frac{d}{dx}\left(h^3\frac{dp}{dx}\right) + 6\mu U_0\frac{dh}{dx} = 0,$$

where  $\mu$  is the (constant) coefficient of viscosity of the fluid.

Write down the equation satisfied by  $\frac{dp}{dx}$ . Explain how this equation can be integrated to find p. If  $h(x) = \frac{h_1}{L} \sqrt{L^2 + x^2}$  for  $-L \le x \le L$  ( $h_1$ , L constants), express the solution for p in terms of integrals. Any integration constants that appear in the solution should be evaluated explicitly.

**4.** Consider the steady flow  $U_0$  of a viscous fluid, with kinematic viscosity  $\nu$ , past a body of typical size L. Define the Reynolds number Re for this flow and discuss briefly the significance of Re in the flow of a viscous fluid.

Explain why it is necessary to introduce a boundary layer to determine the flow past a solid body at large Reynolds number. Use a scaling argument to estimate the thickness of this boundary-layer region in terms of the Reynolds number.

The boundary-layer equations for two-dimensional, steady flow near a solid boundary (y = 0) are

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0,$$

$$u\frac{\partial u}{\partial x} + v\frac{\partial u}{\partial y} = U\frac{dU}{dx} + v\frac{\partial^2 u}{\partial y^2},$$

where U(x) is the outer flow and (u, v) are the velocity components in the x and y directions respectively.

For the flow past a flat plate, where  $U(x) = U_0$ , a constant, show that the equations have a similarity solution in the form

$$\psi = (2\nu U_0 x)^{1/2} f(\eta), \text{ where } \eta = y \left(\frac{U_0}{2\nu x}\right)^{1/2},$$

MATUOOTZ

and where  $\psi$  is the stream function, defined in the usual way. Determine the equation and the boundary conditions satisfied by  $f(\eta)$ .

5. Starting with the boundary-layer equations for steady, two-dimensional flow with an outer flow U(x), derive the momentum integral equation

$$\frac{\tau_w}{\rho U^2} = \frac{d\delta_2}{dx} + \left(\frac{2\delta_2 + \delta_1}{U}\right) \frac{dU}{dx}$$

identifying the various terms in the equation.

Explain why the velocity profile

$$u = \begin{cases} U(x)(2\eta - \eta^2) & (0 \le \eta \le 1), \\ U(x) & (\eta > 1) \end{cases}$$

where  $\eta = y/\delta$ , is a reasonable one to use for an approximate solution of the boundarylayer equations using the momentum integral equation. Derive an equation for  $\delta(x)$  from the momentum integral equation and use this to show that

$$\delta(x)^{2} = \frac{30\nu}{U(x)^{9}} \int_{0}^{x} U(s)^{8} ds.$$

Thus obtain an approximate value for the skin friction for a constant free stream  $U_0$ .

END